The Effect of Boundary Air Flow on Premixed Stationary Burning Velocity

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Abstract

In this study the effect of boundary condition on premixed methane-air stationary Bunsen flames has been experimentally investigated. Laminar burning velocity is calculated by the concentrated cons method (CCM) and Schlieren photography technique, under the effect of laminar boundary horizontal stream air bulk with a range of air flow speeds. The experimental results have shown that the effect of these boundary conditions in general is small on calculated burning velocity if air flow speed is around between (0-50 cm/s). So, it is suggested that this effect be neglected. This effect so characterized increases at/around Stoichiometric ratio flames because of the increase of temperature difference between flame and boundary. The results of the experimental findings were compared with the latest published work and showed a good agreement with it, with a maximum discrepancy of ($\pm 2.5\%$ at j = 1.1).

Keywords: Bunsen flame, laminar, Burning velocity, boundary effect

تأثير جريان الهواء المحيط باللهب الساكن مسبق الخلط على سرعة الاحتراق الخلاصة

Introduction

The Bunsen flame is one of the oldest known examples of stabilized premixed combustion, and it is most important of the simplest examples of multidimensional combustion field. As it have been used for decades to measure laminar flame speed (burning velocity S_u) [1, 2, 3, 4, 5 & 6]. So the previous studies showed that the burning velocity occurs approximately between (37- 43 cm/s) for methane-air mixtures (standard value) [7].

Assumptions have to be made about the velocity profiles of the

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gases released from the burner. The nature of flame shapes was already reviewed by [8]. The approximated shape calculation of Bunsen flame for LPG (liquefaction petroleum gas) is achieved by using concentrated cons method (CCM), suggested by [9].

$$S_u = 2U_{mix} \left(\frac{R_t}{R}\right) / \left(C_1 + C_2\right) \dots (1)$$

Where S_u burning velocity in cm/s, U_{mix} unburned speed of gas mixture, $\left(\frac{R_t}{R}\right)/(C_1 + C_2)$ are

geometrical dimension of flame surface and burner tube. Figure (1) illustrated these dimension.

Laminar burning velocity values and flame stability criteria such as flashback and blow-off ranges are generally affected by geometrical conditions or aerodynamic conditions [9,10,11,12 & 13] and physiochemical conditions such as temperature, pressure and chemical construction [8, 13, 15, 16, 17, 18 & 19]. But in Bunsen flames the geometrical conditions or aerodynamic conditions are considered important factors for flame speed, in addition to the physiochemical conditions, because the burning velocities are calculated from volumetric flow rates and the surface area of flames.

Influence of boundary conditions in large eddy simulation LES of premixed combustion instabilities is studies by [20], the electric field effect on gas temperature, and radiative heat flux and flame speed of premixed CH4/O2/N2 flames is investigated by [21]. Experiments were performed on laminar Bunsen flames (Re<2200) of lean to rich mixture composition

The present study is introducing external air affecting on burning velocity. This condition depends on boundary horizontal stream air bulk that has an effect on Bunsen flame shapes, or in other words (the effect of force convection as over all phenomena), that relies on the literature concerned.

Flame images are captured by using Schlieren photography technique. The advantages of this technique are illustrated in [9, 22, 22 & 24].

Experimental Set-up

The Bunsen burner designed here depends on the requirements of the burner design as described in [8]. The burner tube which is a copper tube with 100 cm length and 1.2 cm inside diameter has been used to obtain a wide range of equivalence ratio (f) and a range of flame stability with approximated zero external air speed. The details of the premixed chamber are illustrated in figure (2).

Fully developing air stream flow that is cutting across with the study phenomenon is achieved by designing of a fully developing channel [24], as illustrated in figure (3). Air flow is obtained by using $60 \times 60 \times 20$ 12V DC Flat Fan with a maximum airflow (23 CFM, $0.0283(m^3/m)$.

The fan selection and specification is depended on figure (4). The regulation of motor speed is achieved by using variable power supply dc source. Air bulk flow speed is measured by digital velocity-meter connected to the Pitot tube that is located near the study phenomenon [26] **. Fan setup and depended variables are show in table (1).

Another setup is achieved to obtained stable flame are show in tables (2 & 3)

Flow meters calibration formulas are

$$V_a = 3.9304 \times a_R + 7.7035...(m^3 / s)$$
... (2)
$$V_f = 0.51158 \times f_R + 2.2443....(m^3 / s)$$
... (3)

$$j = \frac{(m_f / m_a)}{(m_f / m_a)_{st}} = \frac{(V_f / V_a)}{(V_f / V_a)_{st}}$$
$$j = \frac{AFR_{st}}{AFR}$$

Calculation of AFRst (stoichiometric Air to Fuel Ratio)

The stoichiometric quantity of oxidizer is just that amount that is necessary to completely burn a quantity of fuel. The stoichiometric AFR is calculated by balancing C, H, and O atoms in the combustion reaction. Complete combustion of a general hydrocarbon with atmospheric air is written as

$$C_x H_y + a(O_2 + \frac{0.79}{0.21}N_2) \rightarrow$$

 $\rightarrow x CO_2 + \frac{y}{2}H_2O + \frac{a0.79}{0.21}N_2$

By this chemical formula we obtained on table (3), and by floating metering we get the table (4).

Schlieren images of Bunsen flames are captured by commercial camera with 100 ISO Konica films with the camera being adjusted at (F/5.8 and 250 ASA). Schlieren images are saved as JPG extension files with 300 dpi resolution. The contrast and brightness of the images are treated by Adobe Photoshop and dimensions of Schlieren images are captured by AutoCAD to obtain more The sample of captured details. dimension and relation with j and U_a^{flame} are show in table (5), where reading dimension technique is

Actual dimension = measuring dimension \times scale ratio (the scale ratio for Schlieren images is equal to 1.41361256) therefore,

$$H = 1.41361256.H_m...(cm).....(4)$$

$$d = 1.41361256.d_m...(cm).....(5)$$

$$h_1 = 1.41361256.h_{1m}...(cm)....(6)$$

Results and Conclusions

In this study that is not have a numerical relation between the effecter and burning velocity, but the act of the effecter are experiment as well as over all phenomenon under study. A number of experiments are made at room temperature $25 C^{\circ}$ as details in sec.2. Calculated laminar burning velocity is achieved by eq.(1) and the results of these calculation are illustrated in Figures (5) in a wide range of equivalence ratio under variant effect of external The values of calculated air flow. burning velocity are shown that let at the standard value (37-43 cm/s). In otherwise the burning velocity are decreasing with the increase of air flow speed, this state occur duo to increase of the effect of flame quenching resulting from the heat transfer ratio being increased at the flame surface, with this being increased because of force convection - a best approximation of the above is the flow over cylinders, spheres, geometries and other [26]***).

The effect of air flow has been found to be very strong in the range (0.8-1.4) of equivalence ratio as show in shadow reign; this is due to the difference between flame temperature and air flow temperature being larger than the difference at far lean and rich sides. Also it has been shown that if the effect of air flow speed is between (0-50 cm/s), it can be neglected due to the fact that the burning velocity values of methaneair mixtures being in the range specified above in (section 1).

The figure (6) is given comparison between present work and published work depended on ref. [7] to rechecking the present work. And this comparison is illustrating good agreement with published work.

In the future studies we suggested increasing the strong of this effect by using multistage of fans to increasing CFM of system.

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Notes:

**section 4-606, ref [26]

*** Section 3-2 to 3-26, ref [26]

Flo over a flat plate and wedges were classified as laminar or turbulent, depending on the Reynolds number, and correlations for the local and average convective heat transfer coefficients were developed. But flows over cylinders (perpendicular to the axis) and spheres are more complex. In general, the flow over cylinders and spheres may have a laminar boundary layer followed by a turbulent boundary layer

$$q'' = h(T_s - T_{av}) \qquad \operatorname{Re}_d = \frac{\rho U_{\max} d}{\mu} \qquad \operatorname{Nu} = \frac{h_o I}{k} = \frac{qL}{A_s \Delta T k} \qquad \Delta T - T_s - T_e$$
$$\operatorname{Re}_d < 10,000: \qquad \operatorname{Nu}_d = 0.3 + \frac{0.62 \operatorname{Re}_d^{1/2} \operatorname{Pr}^{1/3}}{\left[1 + (0.4/\operatorname{Pr})^{2/3}\right]^{1/4}}$$
$$\underbrace{U_{\infty} T_{\infty}}_{\blacksquare} \qquad \underbrace{T_s}_{\blacksquare}$$

Table (1) Air bulk flow speed depended on $(60mm \times 60mm)$ cores section channel

volt	External air flow rate		Air velocity		
	ft^3/m	cm^3/s	$U_a^{fan} cm/s$	$U_a^{flame} cm/s$	
6.00	12.34	20566.67	571.30	155.493	
4.00	6.32	10533.33	292.59	101.9406	
2.00	3.05	5083.33	141.20	49.53184	
Where <i>volt</i> (voltage regulator), U_a^{fan} (at the fan), U_a^{flame} (at the flame)					

Table (2) General data of mixing

a_{R}	f_R	$T_a(C^{0})$	C_n	H_n
22	10	298	1	4
22	11	298	1	4
22	12	298	1	4
22	13	298	1	4
22	14	298	1	4
22	15	298	1	4
22	17	298	1	4
22	19	298	1	4

22	21	298	1	4
22	23	298	1	4
22	25	298	1	4

Where a_R (air mixed reading), f_R (fuel mixed reading), $T_a(C^{\bullet})$ (air temperature), C_n (number of carbon atoms in fuel), H_n (number of hydrogen atoms in fuel)

xCO ₂	xH_2O	xN_2	<i>xO</i> ₂	xCO	
0.07274	0.14548	0.73246	0.04932	0	
0.0774	0.15481	0.72877	0.03902	0	
0.08202	0.16404	0.72513	0.02881	0	
0.08659	0.17319	0.72151	0.01871	0	
0.09112	0.18224	0.71794	0.0087	0	Stoichiometric
0.09309	0.19097	0.71355	0	0.00239	
0.06169	0.20463	0.69305	0	0.04062	
0.03207	0.21752	0.67371	0	0.07669	
0.00406	0.22971	0.65543	0	0.1108]
-0.02247	0.24126	0.63811	0	0.1431	
-0.04763	0.25221	0.62168	0	0.17374]

Table (3) *x*-mole of fuel construction of CH_4

 Table (4) Unburned mixture parameters

f %	AFR	$U_{mix}(cm/s)$	j			
4.1525	23.082	88.224	0.7468			
4.4283	21.582	88.59	0.7987			
4.7026	20.265	88.957	0.85061			
4.9753	19.099	89.323	0.90252			
5.2464	18.061	89.69	0.95443			
5.516	17.129	90.057	1.0063			
6.0506	15.527 90.792 1.1102					
6.5792	5.5792 14.199 91.529 1.214					
7.1019	7.1019 13.081 92.266 1.3178					
7.6188	7.6188 12.125 93.004 1.4216					
8.1299 11.3 93.744 1.5254						
Where $f \%$ (fuel present), f / a (air to fuel						
ratio), $U_{mix}(cm/s)$ (fully develop mixture						
velocity, j (equivalence ratio)						

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j = 0.7468 or 4.1525% fuel					
Reading	Bound	ary air veloc			
s	0.0	49.53184	101.9406	155.493	
$H_m(cm)$	1.7	1.75	1.8	1.8	
$d_m(cm)$	0.82	0.82	0.82	0.820.	
$h_{1m}(cm)$	0.13	0.13	0.13	0.14	
	j =	1.0063 or 5.	516%fuel		
Reading dimensio	Bounda	ry air veloci	ty effect (ext cm/s	ternal effect)	
ns	0.0	49.53184	101.9406	155.493	
$H_m(cm)$	1.12	1.12	1.13	1.13	
$d_m(cm)$	0.74	0.74	0.74	0.74	
$h_{1m}(cm)$	0.12	0.12	0.12	0.13	
j = 1.5254 or $8.1299%$ fuel					
Reading dimensio	Boundary air velocity effect (external effect) cm/s				
ns	0.0	49.53184	101.9406	155.493	
$H_m(cm)$	2.32	2.33	2.33	2.35	
$d_m(cm)$	0.84	0.74	0.74	0.74	
$h_{1m}(cm)$	0.16	0.16	0.16	0.17	

 Table (5)
 Finishing experimental setup



Figure (1) geometrical description of Benson flame surface



Figure (2) Benson premixed fuel chamber detail



Figure (3) Fully developing channel with burner and fan locations



Figure (4) Impeller Flat fan specification (<u>www.sunon.com</u>)



Figure (5) Relation between burning velocity and effected external air bulk



Figure (6) Comparison between present work and published work depended on ref [7]